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Aerothermodynamics and exergy analysis in turbocharger radial turbine

Shyang Maw Lim, Anders Dahlkild, Mihai Mihaescu

KTH Mechanics, Competence Center for Gas Exchange (CCGEx)



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The turbocharger turbine





Laurantzon F, Örlü R, Segalini A, Tillmark N, Alfredsson H. Experimental analysis of turbocharger interaction with a pulsatile flow through time-resolved flow measurements upstream and downstream of the turbine. In: Institution of Mechanical Engineers - 10th International Conference on Turbochargers and Turbocharging [Internet]. 2012. p. 405–15.

Lyttek P, Roclawski H, Böhle M, Gugau M. New Modular Test Rig for Unsteady Performance Assessment of Automotive Turbocharger Turbines. ASME. Turbo Expo: Power for Land, Sea, and Air, Volume 8: Microturbines, Turbochargers and Small Turbomachines; Steam Turbines ():V008T26A017. doi:10.1115/GT2017-64218.



Aerothermodynamic losses?

Romagnoli, A. Manivannan, S. Rajoo, M.S. Chiong, A. Feneley, A. Pesiridis, R.F. Martinez-Botas, "A review of heat transfer in turbochargers," Renew. Sustain. Energy Rev., vol. 79, no. Supplement C, pp. 1442–1460, 2017



Research gaps & goal



Limited insights into the thermo-fluid effects due to heat transfer with the first law of thermodynamic based method

Customary set-up ignore the presence of the upstream exhaust manifold

Limited knowledge about the thermo-fluid physics under on-engine conditions

Goal: Understand the thermo-fluid physics and the turbine performance under realistic engine conditions



Research questions & strategy







Challenges & strategy





Computational model

KTH vetenskap voch konst





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0.2

0.4

 t/t_{cycle}

0.6

[-]

0.8



Thermal BCs



Component	Wall thermal conditions
Exhaust manifold & scroll	T _w = 1173 K
Rotor	Adiabatic
Diffuser & outlet duct	T _w = 929 K

 T_w is the estimated average solid-fluid wall temperature from 1D model





Flow field (1500 rpm)







Flow exergy
$$e_f = [h_t(T_t) - h_o] - T_o[s(T, p) - s_o]$$



$$\dot{S}_{gen,viscous} = \iiint_{V^*(t)} \left[\frac{\mu_{mol} + \mu_{turb}}{T} \Phi \right] dV \qquad \dot{S}_{gen,thermal} = \iiint_{V^*(t)} \left[\frac{k_{mol} + k_{turb}}{T^2} (\nabla T)^2 \right] dV$$



Flow exergy buc	lget	G
$- \oint_{S^*(t)} \rho e_f(\vec{u} - \vec{u}_b) \cdot \vec{n} dS$		Flow exergy change (Inflow - Outflow)
$= \frac{d}{dt} \left[\iiint_{V^*(t)} (\rho e_f) dV \right] - \frac{d}{dt} \left[\iiint_{V^*(t)} p dV \right]$		= Unsteadiness
+ $\oint \left[(\vec{r} \times \vec{f}_{n,st}) \cdot \vec{\Omega} \right] dS$	Physical interpretation	+ Everal transferred by shaft work
$\int \int [(t+y) het f(t)] dt dt$		· Exergy transiented by shalt work
$+ \oint_{S^*(t)} \left[\left(1 - \frac{T_o}{T_b} \right) \right] (\vec{q} \cdot \vec{n}) dS$		+ Exergy gained/lost by heat transfer
$+ T_o(\dot{S}_{gen,viscous} + \dot{S}_{gen,thermal})$		+ Exergy destroyed by irreversibilities





Flow exergy budget (1500 rpm)





Scroll is most affected

Turbine power is relatively unaffected





What are the effects of upstream exhaust manifold on the heat transfer rate and performance of the turbine?

- Fundamental thermo-fluid physics (i.e. secondary flow) in the scroll is governed by exhaust manifold
- Heat transfer rate & aerothermodynamic losses increase
- Turbine power is relatively unaffected







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