



ROYAL INSTITUTE  
OF TECHNOLOGY

# Accurate determination and control of compressor noise

**Raimo Kabral**

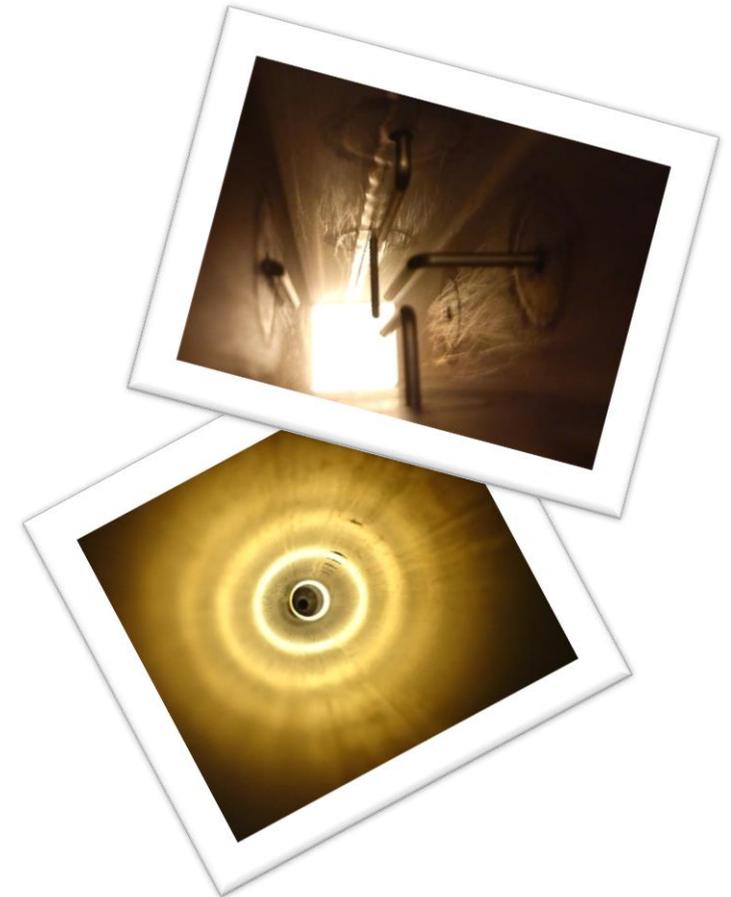
supervised by **Prof. Mats Åbom** and  
**Prof. Hans Bodén**

Competence Center Gas Exchange  
**CCGEx**

”Charging for the future”

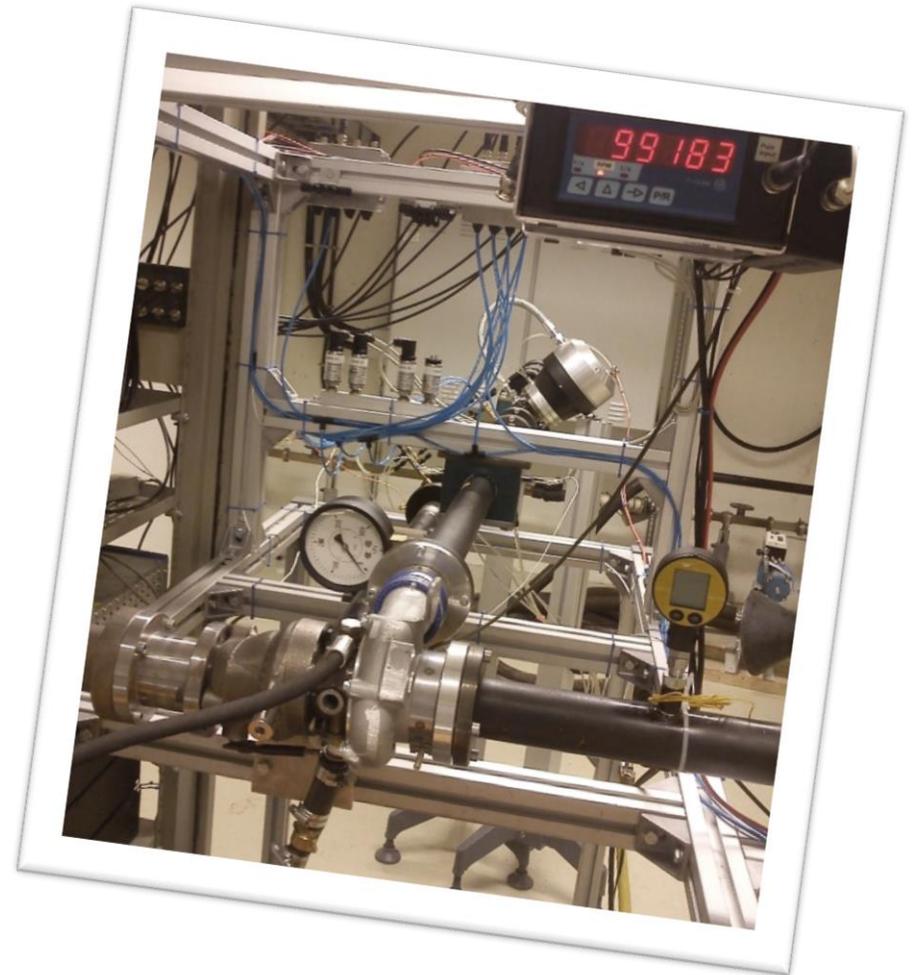
# Layout of the presentation

1. Compressor studied
2. Accurate determination of sound generation
3. Dissipative noise control of the compressor
  - ✓ Concept of compact silencer
  - ✓ Improved FEM model
  - ✓ Modified optimal impedance model

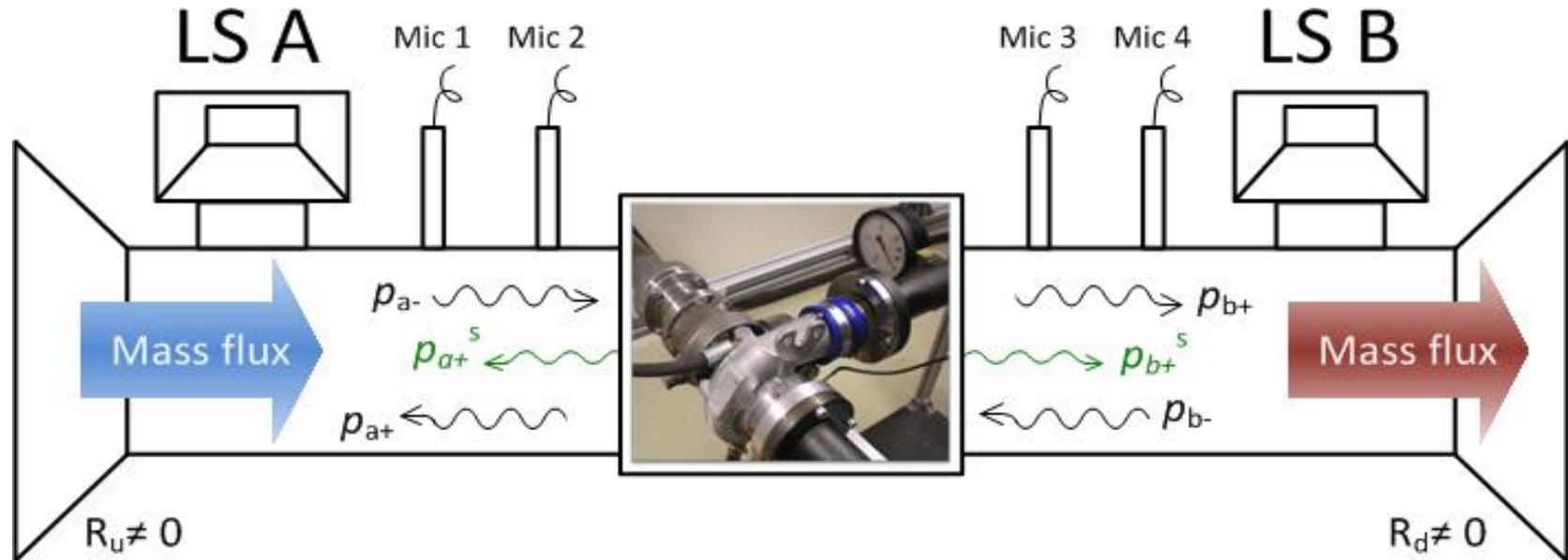


# Compressor studied

- Passenger car turbocharger  
Garrett GT1752 driven by the  
“warm” compressed air feed  
to the turbine.
- Inlet diam. is 44mm.
- Outlet diam. is 42mm.
- The rotor has 6  
(+6 splitter) blades.
- Shaft frequency  
~80...180kRPM – blade pass  
frequency 8...18kHz.



# Reflection-free sound generation

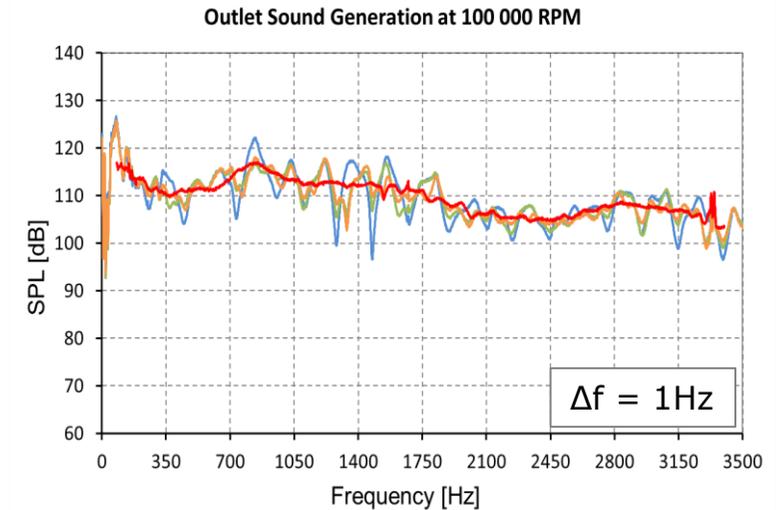
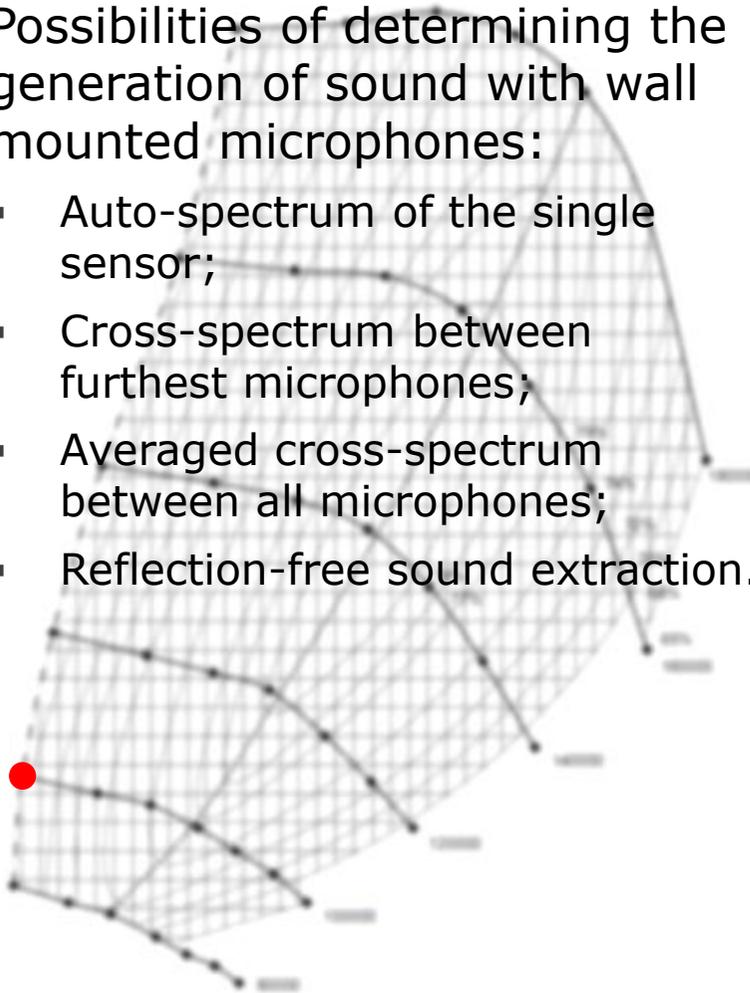


$$\mathbf{p}_+^s = (\mathbf{E} - \mathbf{SR})(\mathbf{E} + \mathbf{R})^{-1} \mathbf{p}$$

$$\mathbf{G}^s = \mathbf{p}_s (\mathbf{p}'_s)^\dagger = \begin{bmatrix} G_{p_a^s p_a^s} & G_{p_b^s p_a^s} \\ G_{p_a^s p_b^s} & G_{p_b^s p_b^s} \end{bmatrix}$$

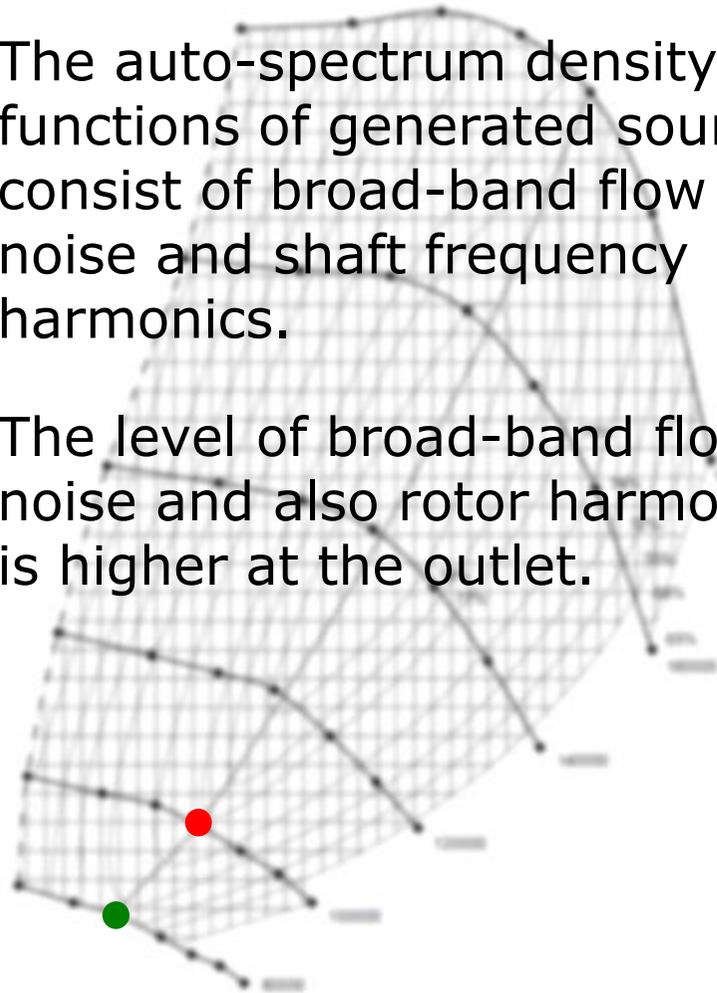
# Sound generation of the compressor

- Possibilities of determining the generation of sound with wall mounted microphones:
  - Auto-spectrum of the single sensor;
  - Cross-spectrum between furthest microphones;
  - Averaged cross-spectrum between all microphones;
  - Reflection-free sound extraction.

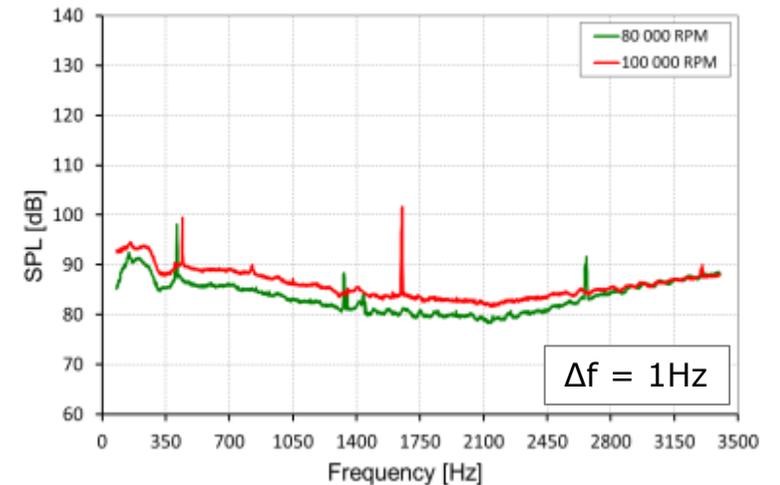


# Results: Sound generation

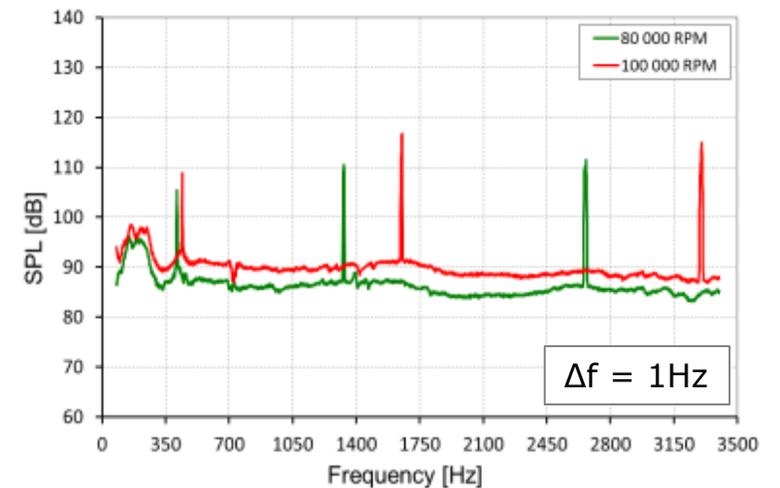
- The auto-spectrum density functions of generated sound consist of broad-band flow noise and shaft frequency harmonics.
- The level of broad-band flow noise and also rotor harmonics is higher at the outlet.



Inlet Sound Generation at Peak Efficiency Line

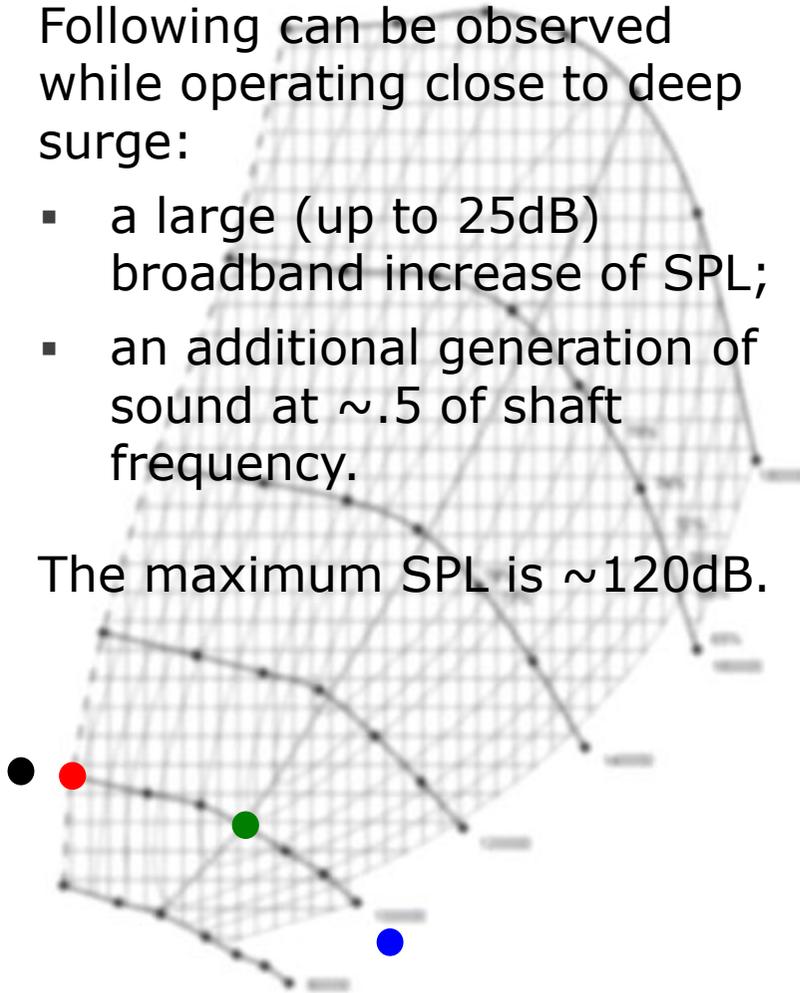


Outlet Sound Generation at Peak Efficiency Line

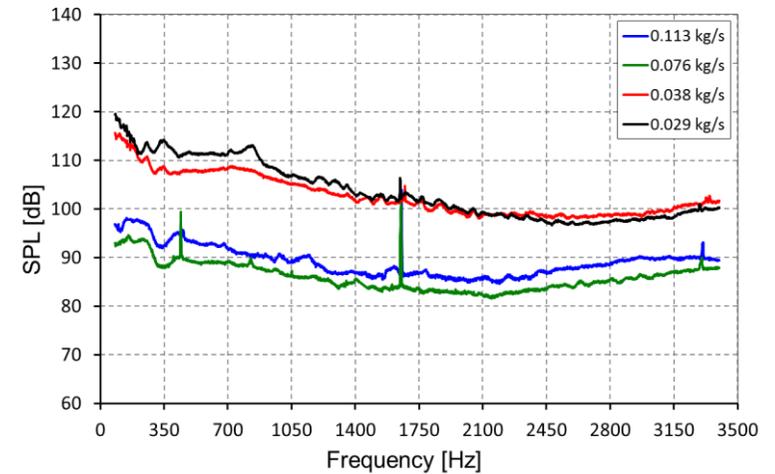


# Sound generation of the compressor

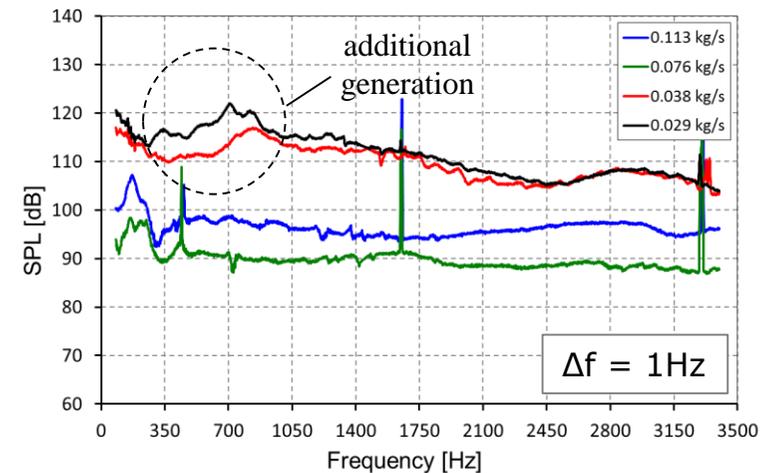
- Following can be observed while operating close to deep surge:
  - a large (up to 25dB) broadband increase of SPL;
  - an additional generation of sound at  $\sim .5$  of shaft frequency.
- The maximum SPL is  $\sim 120$ dB.



Inlet Sound Generation at 100 000 RPM

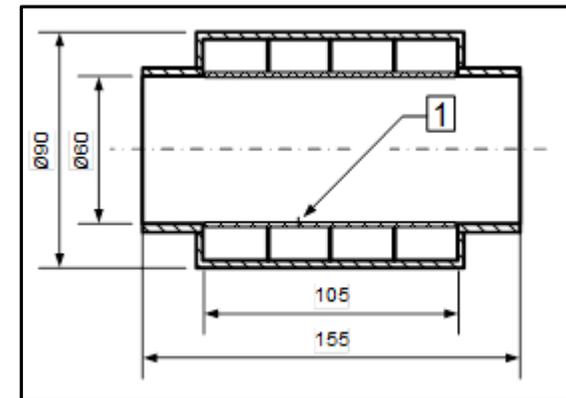
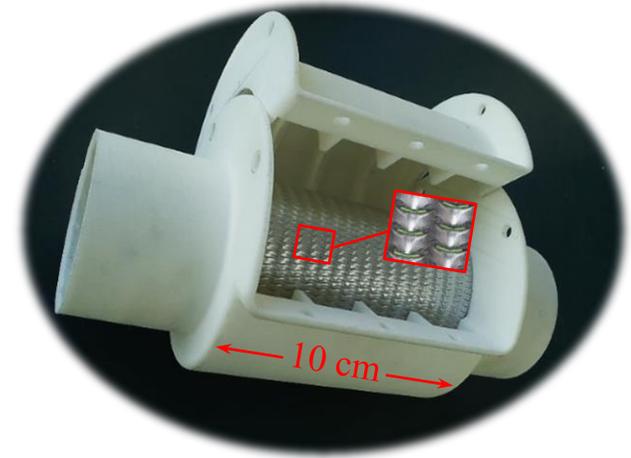


Outlet Sound Generation at 100 000 RPM



# Concept of compact silencer

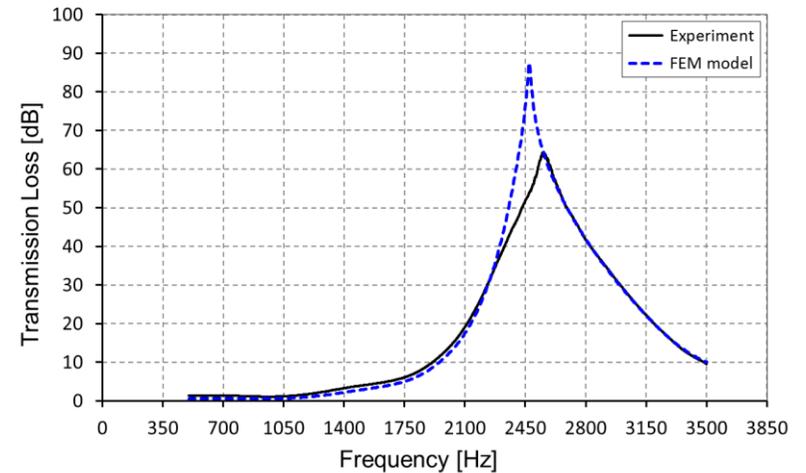
- Lightweight and compact noise control solution for flow duct applications (e.g. compressors).
- Consist of straight-flow channel with included acoustic **resistance** and **locally reacting cavity**.
- The prototype silencer employs the custom Acustimet™ MPP.
- The acoustic performance is controlled by the acoustic impedance of the locally reacting surface.



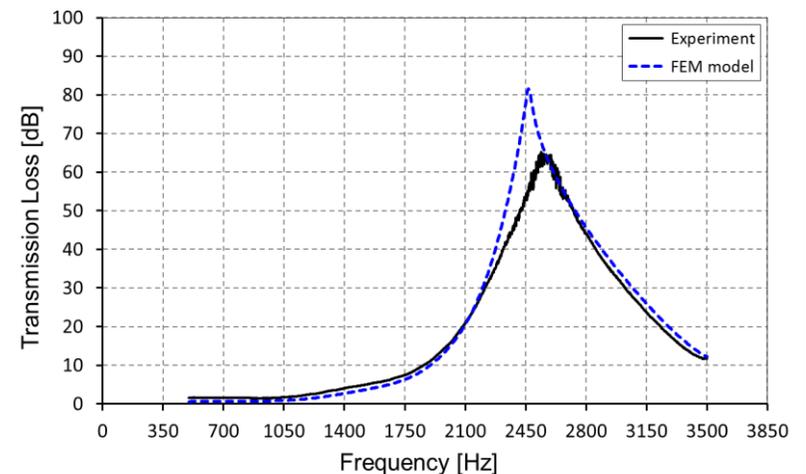
# Improved FEM model

- Improvements are:
  - ✓ Full 3D → 2D with axisymmetric boundary condition.
  - ✓ Inhomogeneous Helmholtz equation → convective wave equation (“plug flow”).
  - ✓ Modified grazing flow models for the acoustic transfer impedance of the Acustimet™ micro-perforated panels.
  - ✓ Cavity baffles included in the geometry.
- The model can predict the acoustic performance reasonably well.

Transmission Loss in absence of flow



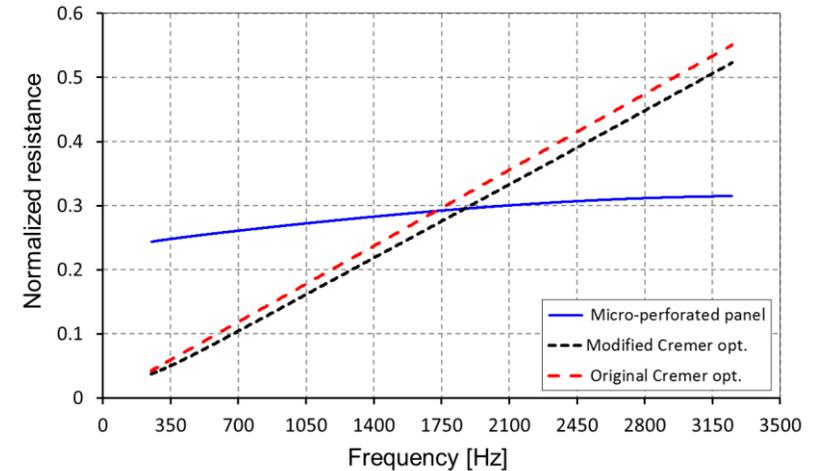
Transmission Loss for Mach number 0.042



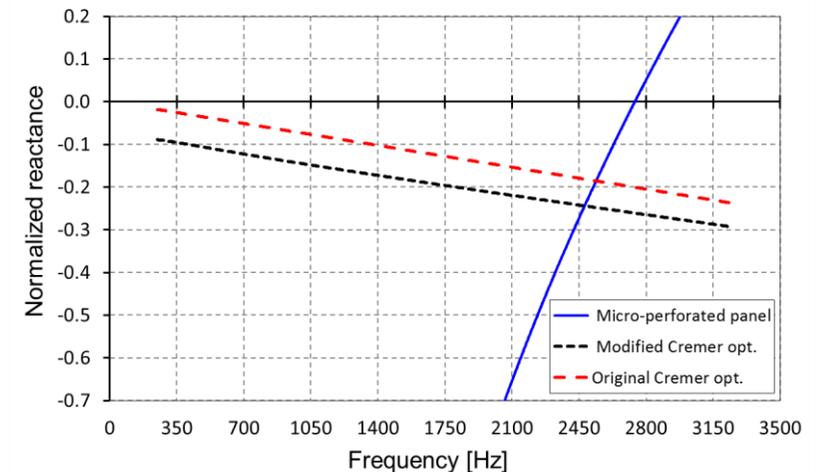
# Modified optimal impedance model

- The assumption of very high frequency (i.e. well cut-on acoustic modes) is eliminated.
- In order to obtain modified Cremer impedance, the boundary condition equation has to be solved numerically.
- Difference in the optimal impedance result will grow with **mean flow Mach number**.

Acoustic resistance for Mach number 0.042

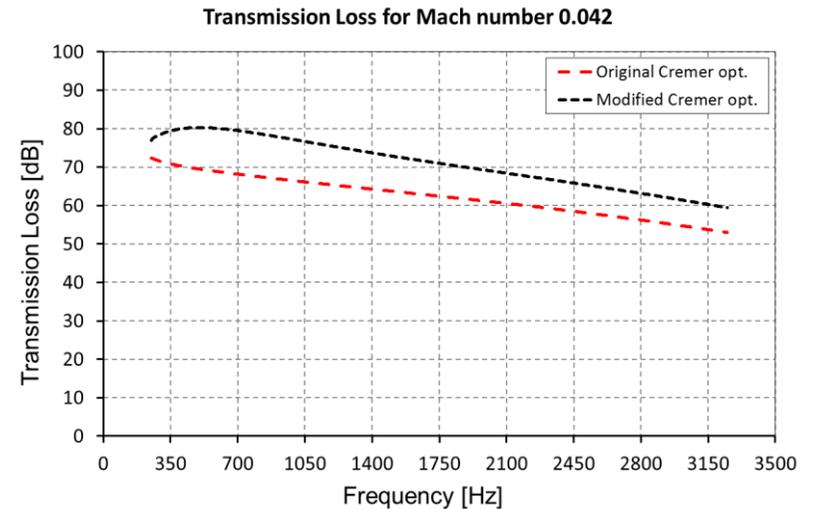


Acoustic reactance for Mach number 0.042



# Modified optimal impedance model

- The modified Cremer impedance model will result in the higher dissipation and TL.



The work is also supported by the “*FlowAirS*” initial training network of European Commission FP7 Marie Curie Actions ([www.flowairs.eu](http://www.flowairs.eu)).

*FlowAirS*



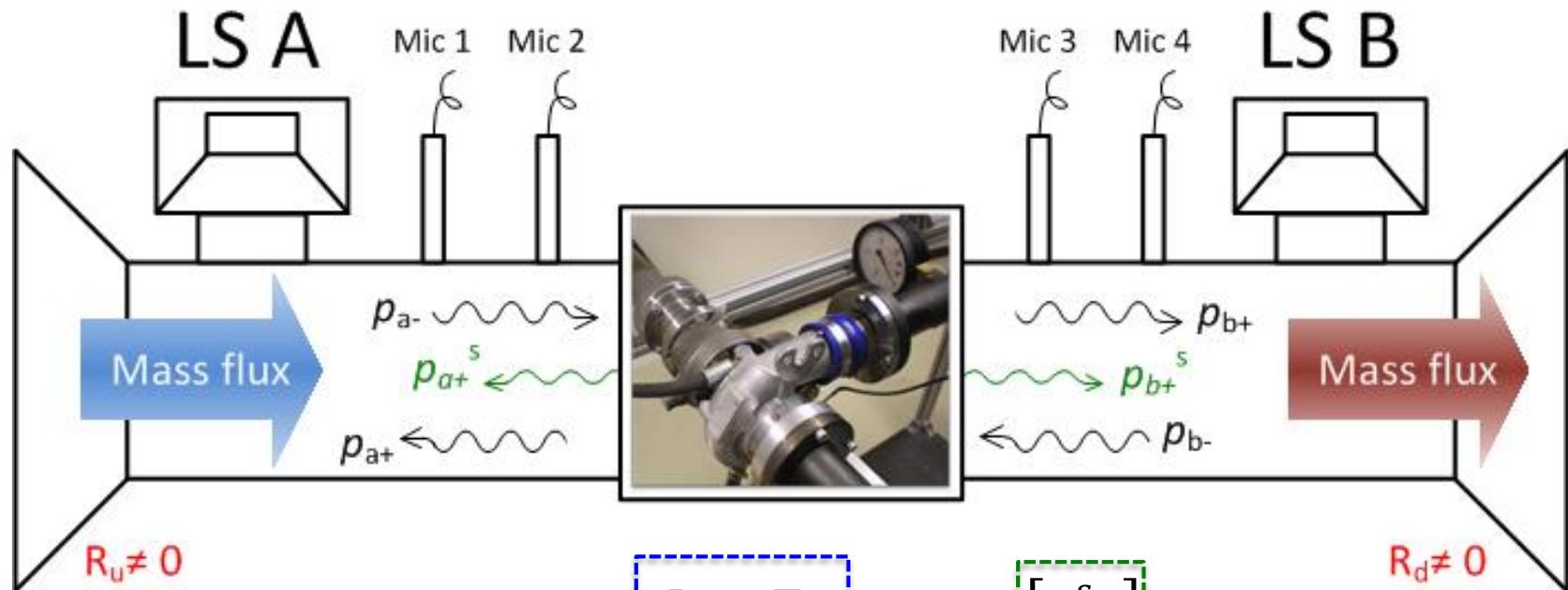
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Thank you for your attention.

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# Acoustic 2-port formulation

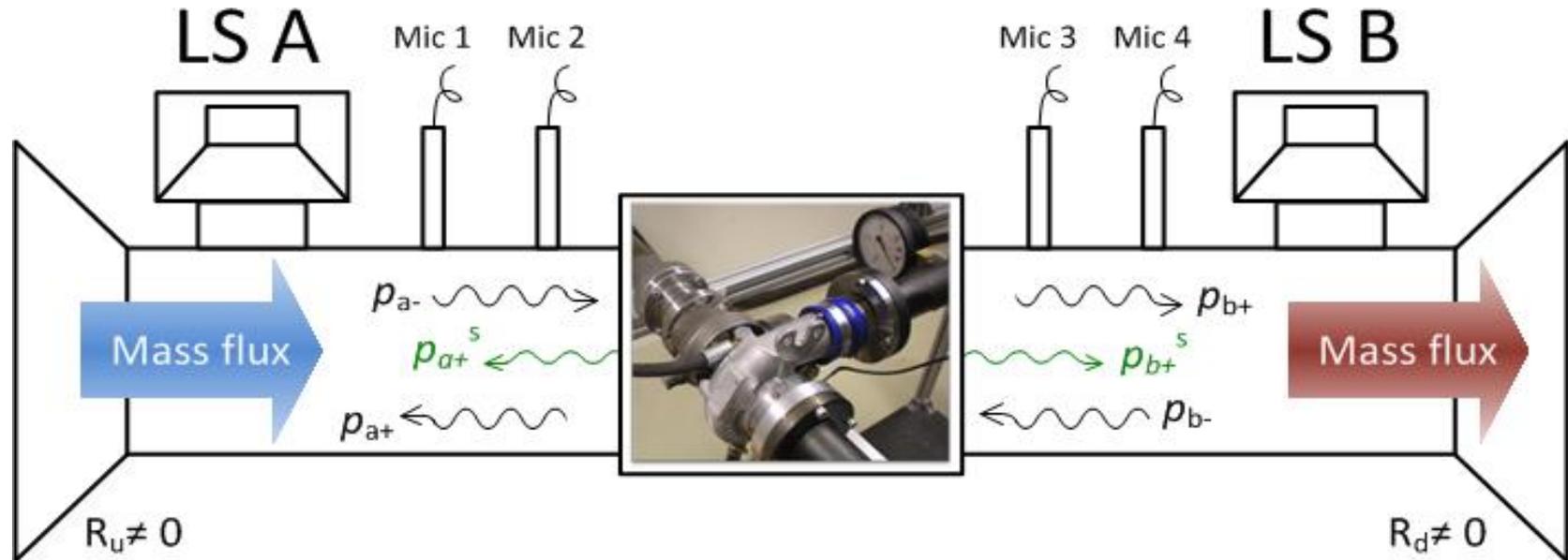


$$\begin{bmatrix} p_{a+} \\ p_{b+} \end{bmatrix} = \begin{bmatrix} R_a & T_b \\ T_a & R_b \end{bmatrix} \begin{bmatrix} p_{a-} \\ p_{b-} \end{bmatrix} + \begin{bmatrix} p_{a+}^s \\ p_{b+}^s \end{bmatrix}$$

- The acoustical performance of a flow duct element is determined by the full 2-port model which consists both the **passive** and the **active** parts.

$$f_c = \frac{1.814 \cdot c}{\pi d}$$

# Reflection-free sound generation

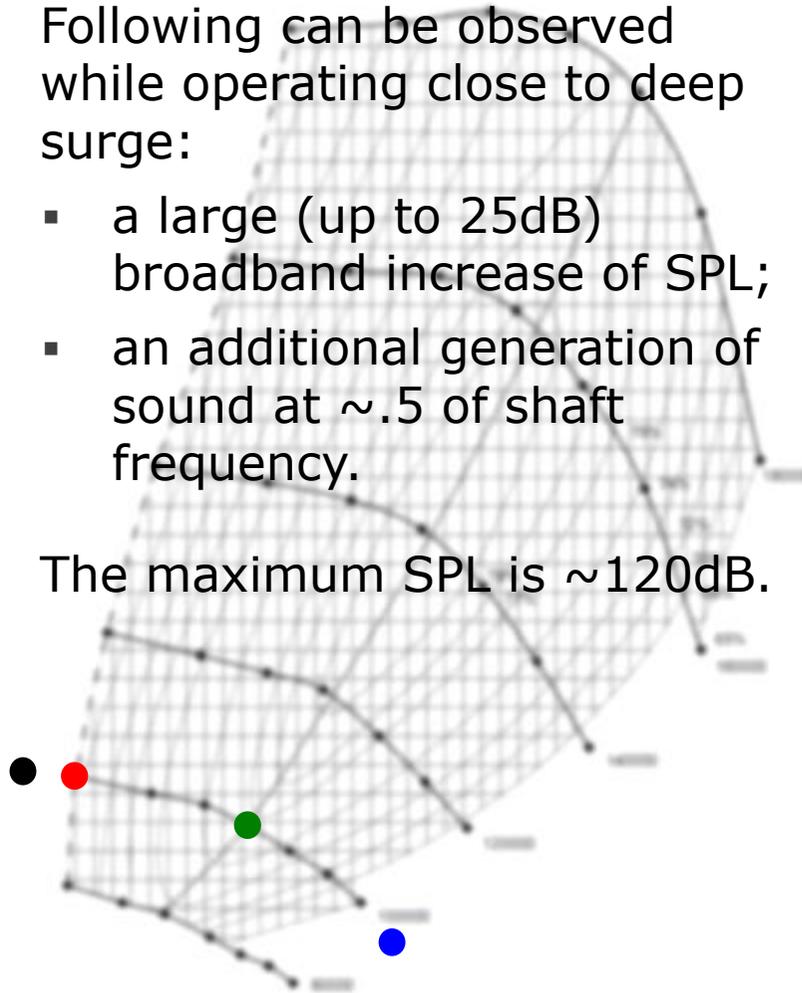


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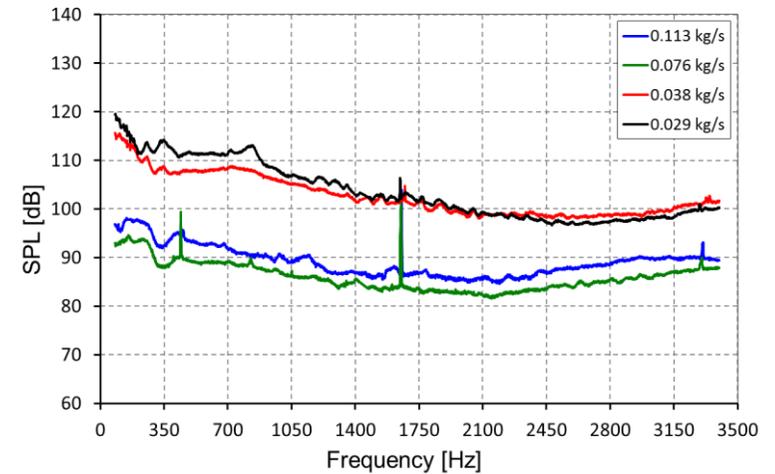
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# Sound generation of the compressor

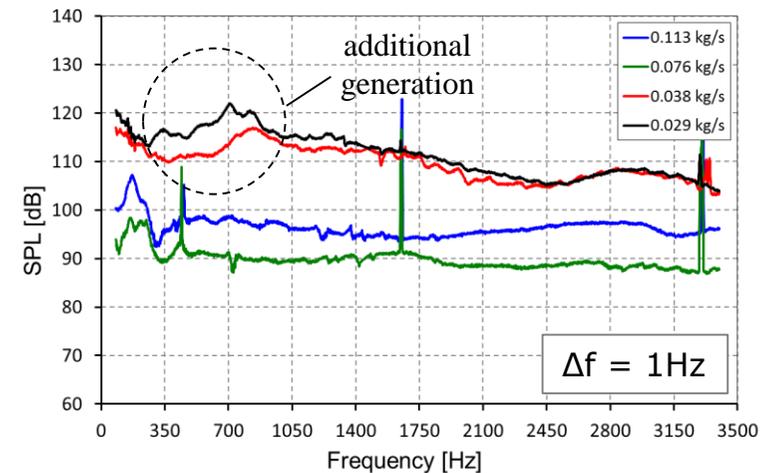
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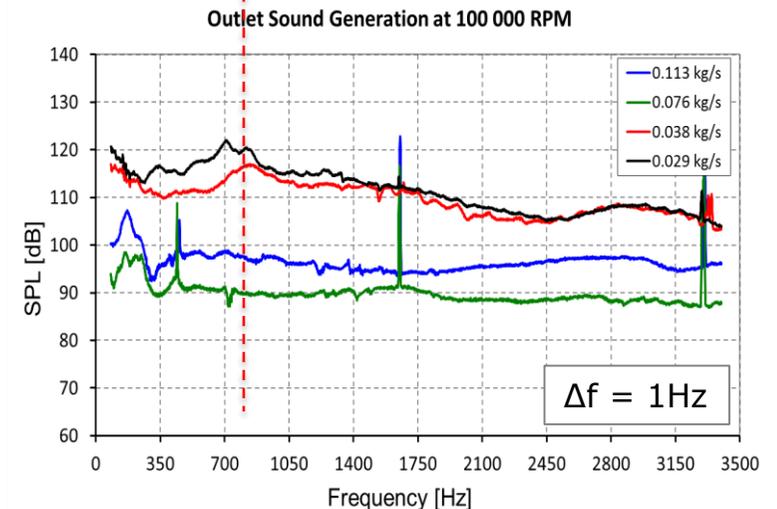
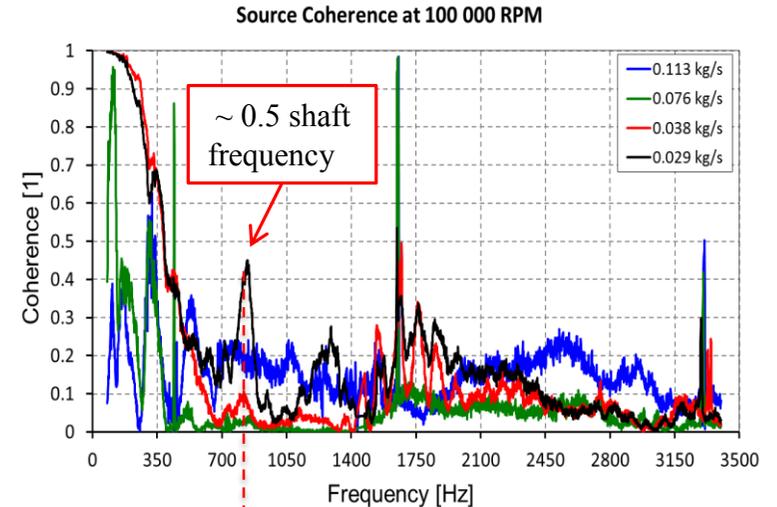
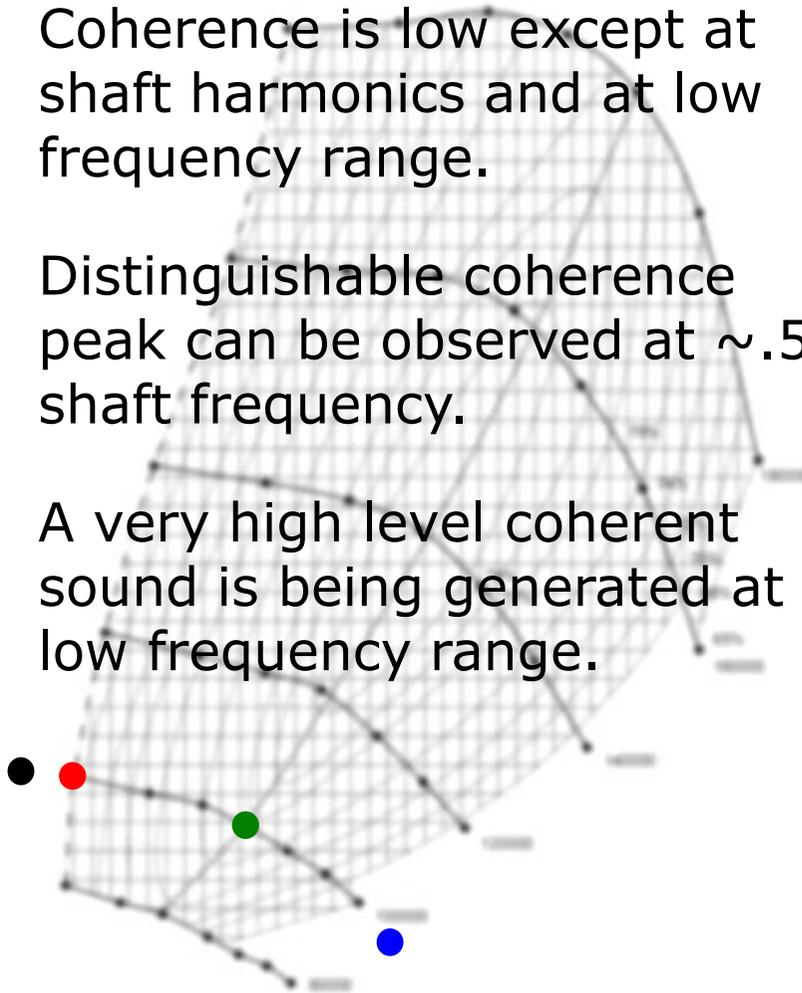


Outlet Sound Generation at 100 000 RPM



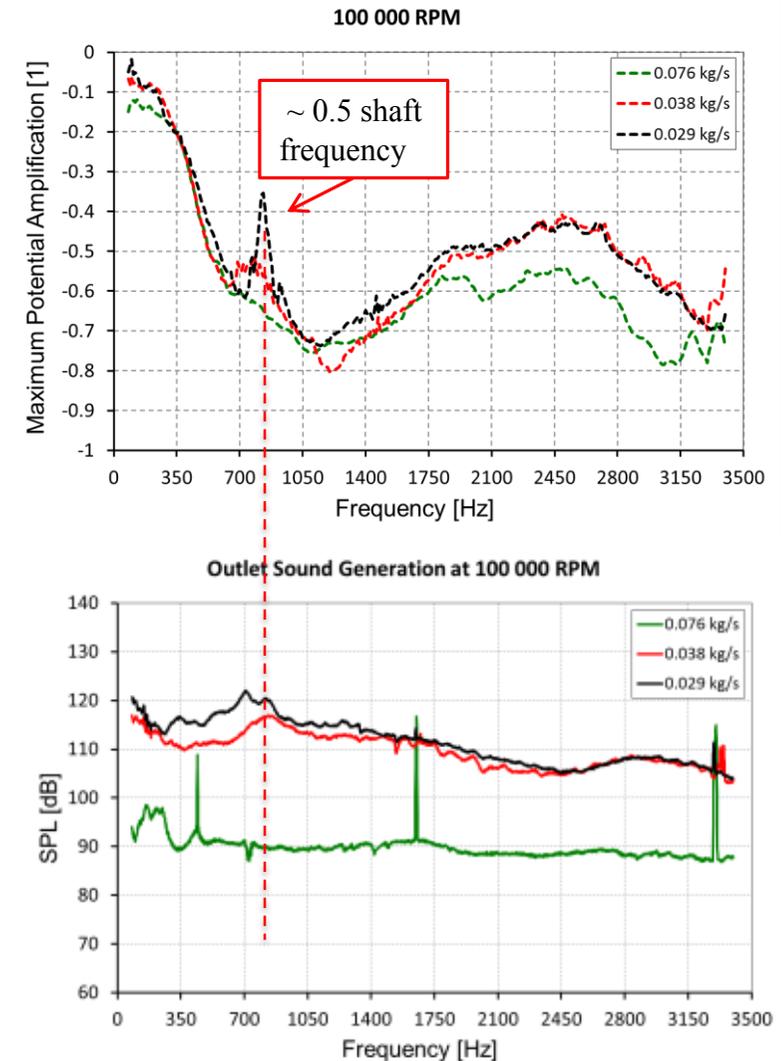
# Compactness of a source

- Coherence is low except at shaft harmonics and at low frequency range.
- Distinguishable coherence peak can be observed at  $\sim .5$  shaft frequency.
- A very high level coherent sound is being generated at low frequency range.



# Surge initiation by acoustic field?

- In case of reflective boundaries, the generated sound is sent back to the compressor.
- At  $\sim 0.5$  of shaft frequency the incident sound energy will be dissipated regardless the local amplification.
- At very low frequency the generated sound energy can accumulate in the system
  - and potentially have a significant effect to flow conditions.



# Surge initiation by acoustic field?

- Negligible dissipation in low frequency range strongly depending on the incident sound field.
- The outlet excitation amplitude becomes dominant in cases of locally low dissipation.

